THE NATURE OF NEAR-EARTH ASTEROIDS FROM THE STUDY OF THEIR THERMAL INFRARED EMISSION

Dissertation zur Erlangung des Grades

eines Doktors der Naturwissenschaften

vorgelegt von

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> und eingereicht im Fachbereich Geowissenschaften der Freien Universität Berlin

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Tag der Disputation: 04. Nov. 2004

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that there are two curves for each value of the thermal parameter Θ : continuous curves refer to those η -values derived by observing the morning side of the asteroid, whereas dasheddotted curves indicate those η -values obtained observing the afternoon side. Curves obtained for Θ =0.025 and 0.13 are not plotted since the derived η -values are constant with phase angle and their values between 1 and 1.05. The dotted black curve represents the expected η values for an FRM-like ($\Theta \rightarrow \infty$) asteroid.

- Fig. 6.7 NEATM derived η parameter as a function of the phase angle and macroscopic surface roughness $\overline{\theta}$. The sun and the observer are in the equatorial plane of the synthetic asteroid. The thermal parameter Θ is equal to 0. Different colors are used for different values of $\overline{\theta}$: η values derived for $\overline{\theta} = 58^{\circ}$ are coded with black color; those obtained for $\overline{\theta} = 36^{\circ}$ are coded with red; for $\overline{\theta} = 20^{\circ}$ with green and for $\overline{\theta} = 10^{\circ}$ with blue.
- Fig. 6.8 Continuous line: diurnal temperature profiles for an equatorial tile of an object with subsolar latitude equal to zero. Dashed-dotted line: diurnal temperature profiles for one of the four tiles on the floor of an equatorial crater with opening angle equals to 45° (a) and with opening angle equals to 90° (hemispherical crater).
- Fig. 6.9 Combined effects of thermal inertia, rotation rate and surface roughness on the theoretical dependence of the NEATM η-value with the phase angle.
- Fig. 6.10 Combined effects of thermal inertia, rotation rate and surface roughness on the theoretical dependence of the NEATM η-value with the phase angle.
- Fig. 6.11 Verification of the hypothesis that η -values derived for asteroids observed from randomly oriented directions are limited by the "morning" curve M and the curve of zero thermal inertia N. The thermophysical model was run for three values of the sub-solar latitude B_{ss}. Crosses represent those η -values derived for asteroids with B_{ss}=0°, asterisks for asteroids with B_{ss}=30° and diamonds for B_{ss}=60°. Note how η -values collapse to the curve of zero thermal inertia as B_{ss} approaches 90°. Following our notation M is the curve with $\eta = \eta(-|\alpha|, \Theta, \overline{\theta})$, A that with $\eta = \eta(|\alpha|, \Theta, \overline{\theta})$ and N that with $\eta = \eta(|\alpha|, \Theta = 0, \overline{\theta})$.
- Fig. 6.12 Limiting curves which do not fit properly the observed η -values. See Fig. 6.13 caption for a description of the symbols.
- Fig. 6.13 Limiting curves which DO fit the observed η -values. The values of the Θ and the $\overline{\theta}$ parameter used to draw the curves are shown on the upper left side of each plot. For each value
- Fig. 6.14 Limiting η - α curves to fit observed η -values of 5381 Sekmeth. Those curves were calculated for Θ =4.4 and $\overline{\theta}$ =36°
- Fig. 6.15 Limiting η - α curves to fit observed η -values of 433 Eros. Those curves were calculated for Θ =1.0 and $\overline{\theta}$ =20°
- Fig. 6.16 Section of the STM relative albedo error function i.e. $(p_{V_STM}(\alpha,\Theta, \overline{\theta})-p_{V_TM})/p_{V_TM} \times 100$ at constant value of $\overline{\theta}$. The refined STM of Lebosfky and Spencer (1989) was used with constant η =0.756 and β_E =0.01 magnitude per degree. The function was numerically evaluated on a grid of ten degree of step size in α and at Θ =[0.13, 0.25, 0.40, 0.50, 1.00, 2.00, 4.60, 12.70, 25.5].
- Fig. 6.17 Section of the STM relative albedo error function i.e. $(p_{V_STM}(\alpha,\Theta, \overline{\theta})-p_{V_TM})/p_{V_TM} \times 100$ at constant value of $\overline{\theta}$. In contrast to Fig. 6.16, here η is constant but equal to 0.95 and $\beta_E=0.015$ magnitude per degree, as described in section 5.6. The function was numerically
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evaluated on a grid of ten degree of step size in α and at $\Theta = [0.13, 0.25, 0.40, 0.50, 1.00, 2.00, 0.50$ 4.60, 12.70, 25.5].

- Fig. 6.18 Section of the NEATM relative albedo error function i.e. $(p_{V_{NEATM}}(\alpha, \Theta, \overline{\theta}) p_{V_{TM}})/(\alpha, \Theta, \overline{\theta}) p_{V$ $p_{V TM} \times 100$ at six different constant values of $\overline{\theta}$. The function was numerically evaluated on a grid of ten degree of step size in α and at Θ =[0.13, 0.25, 0.40, 0.50, 1.00, 2.00, 4.60, 12.70, 25.5]
- Fig. 6.19 Distribution of the albedo relative error as a function of the η -value for asteroid observed at phase angle between -40 and 40 degrees (a) and at phase angle larger than 40° or smaller than -40° .
- Fig B.1 Transmission curves for the filters installed at the LWS (Wirth & Campbell, personal communication, 2000). In the case of the M filter, the curve is the product of the transmission of the filter with that of the atmosphere and the optics of the instrument. 175

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